

Status of ESA's Mars Activities

**MEPAG meeting
Washington DC, 13-14 May 2014**

Rolf de Groot

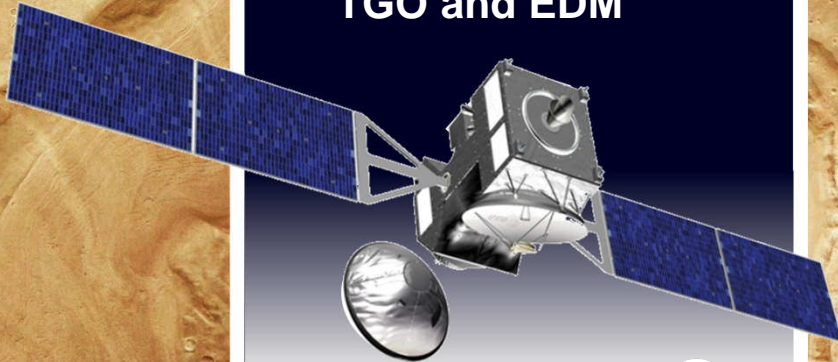
European Space Agency
Robotic Exploration Coordination Office
Science and Robotic Exploration Directorate

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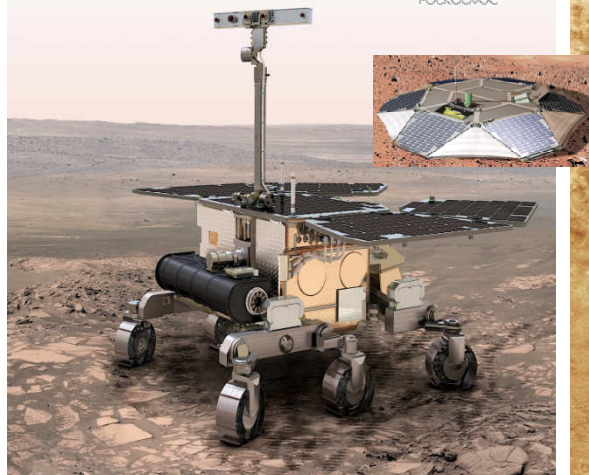


Mars Exploration at ESA

TGO and EDM



ExoMars

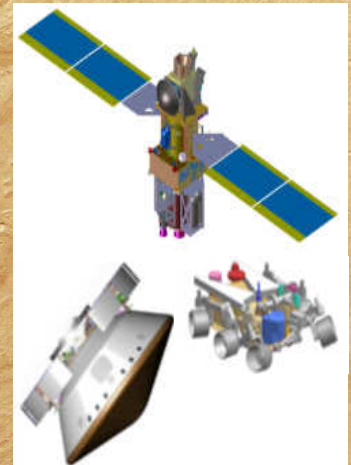
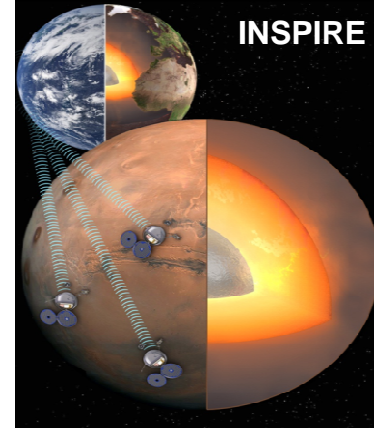


Rover + Platform

PHOOTPRINT



INSPIRE



MSR elements

Mars Express



2003

2030+

Mars Express Mission: Status and near future milestones



- Status:
 - Spacecraft and payload status: very good
 - Fuel estimates enough for a few more years.
 - Mission extension: until end of 2016, 2017-2018 proposal Q4 2014.
 - 29 December 2013: closest ever flyby of Phobos (~ 45 km).
- Near future milestones
 - 19 October 2014: Siding Spring comet flyby (distance ~ 135,000 km)
 - October 2014 onwards: coordinated measurements with the NASA MAVEN mission. Scientific objective: study of the plasma environment and atmospheric escape.
 - October 2016: Mars Express relays data from the ExoMars entry and descent module

ExoMars Programme

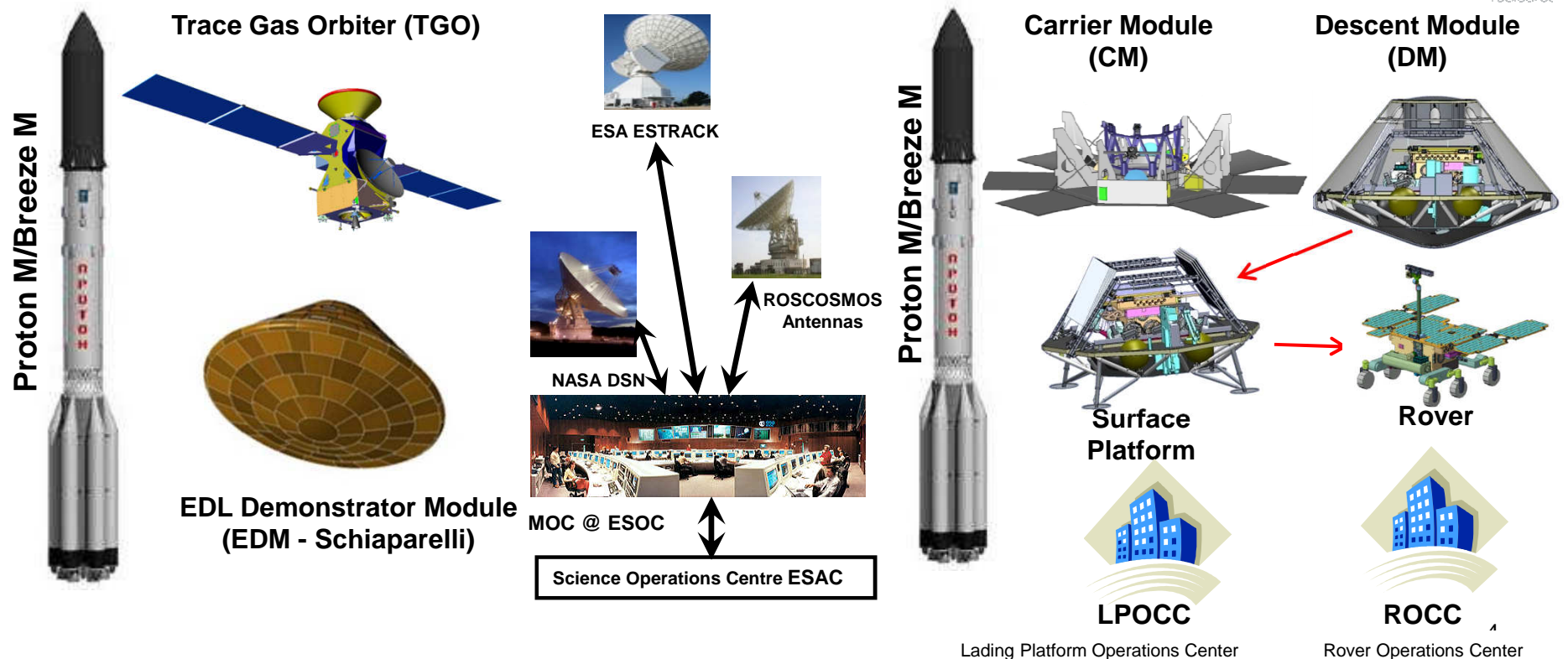
Two missions launched in 2016 and 2018, respectively

- The 2016 flight segment consists of a **Trace Gas Orbiter (TGO)** and an **EDL Demonstrator Module (EDM) - Schiaparelli**
- The 2018 flight segment consists of a **Carrier Module (CM)** and a **Descent Module (DM)** with a **Rover** and a stationary **Surface Platform**

2016 Mission

And

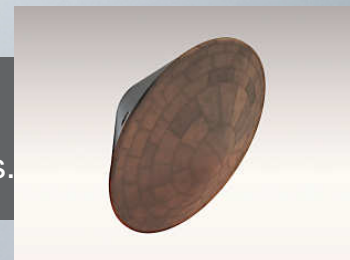
2018 Mission





TECHNOLOGY OBJECTIVE

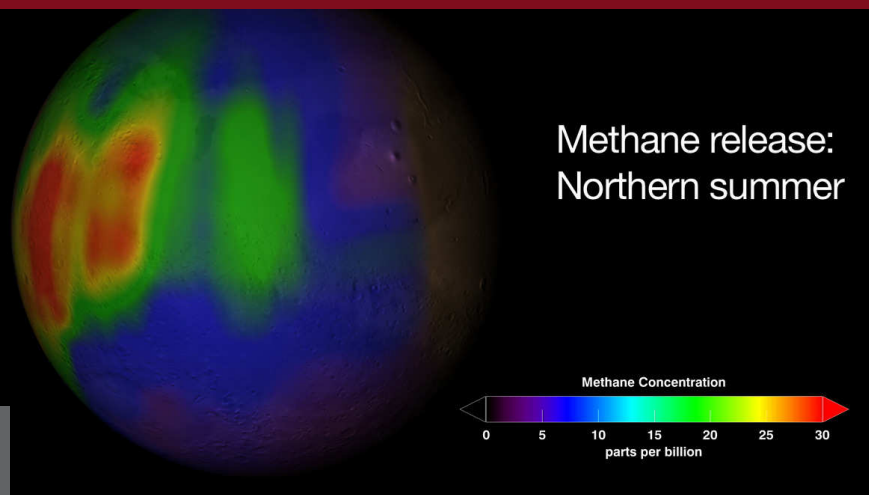
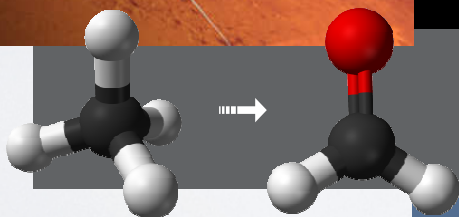
- Entry, Descent, and Landing (EDL) of a payload on the surface of Mars.



2016

SCIENTIFIC OBJECTIVE

- To study Martian atmospheric trace gases and their sources.
- To conduct surface environment measurements.



- Provide data relay services for landed missions until 2022.



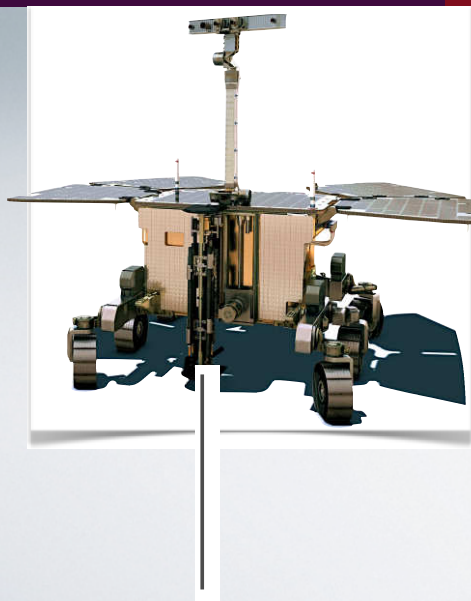
TECHNOLOGY OBJECTIVES

- Surface mobility with a rover (having several kilometres range);
- Access to the subsurface to acquire samples (with a drill, down to 2-m depth);
- Sample acquisition, preparation, distribution, and analysis.
- Qualification of Russian ground-based means for deep-space communication
- Adaptation of Russian on-board computer for deep space missions and ExoMars landed operations
- Development and qualification of throttleable braking engines for prospective planetary landing missions

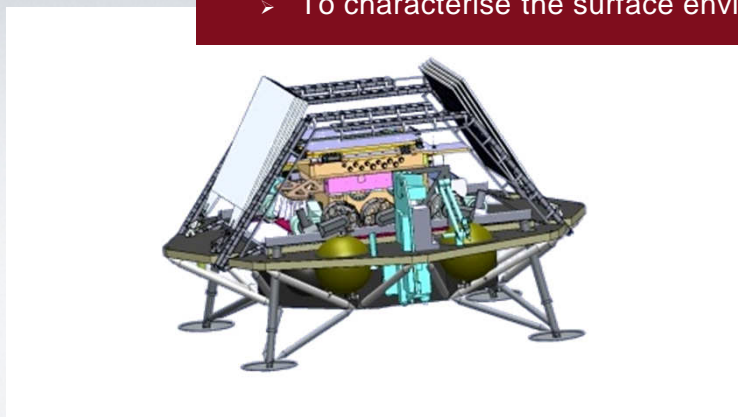
2018

SCIENTIFIC OBJECTIVES

- To search for signs of past and present life on Mars;
- To characterise the water/subsurface environment as a function of depth in the shallow subsurface.



- To characterise the surface environment.

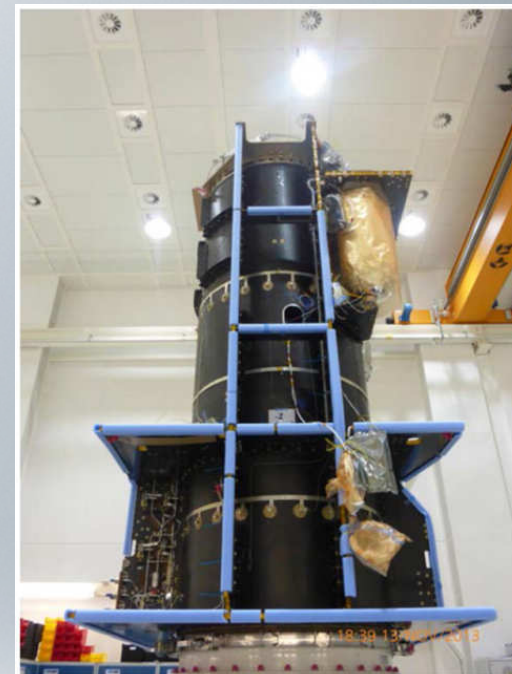


ExoMars programmatic status

- ESA – Roscosmos ExoMars agreement in force (14 March 2013)
- ESA – NASA MoU on ExoMars 2016 in force (29 April 2014)
- Revised Science Management Plan approved (February 2014)

2016 ExoMars Mission (TGO and EDM)

- Contract signed with Industry for full development, launch and operations
- Operations and science ground segment in development
- System Critical Design Review (S-CDR) board meeting on 18-4-2014 concluded no show stoppers but requested further reporting on some issues
- 2016 mission on schedule with limited margins
- Effect of US State Department block on Export Licences still tbd



2018 ExoMars Mission (CM, DM, Rover)

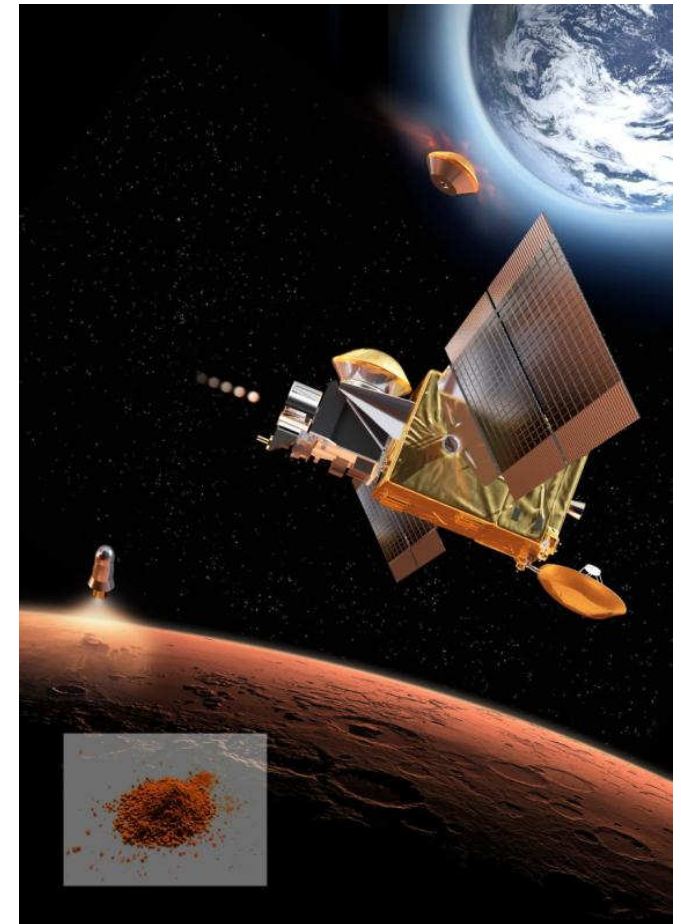
- The 2018 mission is in Phase B (CM, DM), while the Rover is in Advanced CD
- System Requirements Review (SRR) close-out on 17 December 2013
- Landing Site Selection process kicked-off in December 2013 (see *JV presentation*)
- AO for Surface Platform instruments in preparation (Q2 2014 tbc)
- Descent Module Design Project Review ongoing (March-June 2014)
- System Preliminary Design Review (S-PDR) in preparation (May-July 2014)
- Full Industrial Proposal expected in Q3 2014

ESA activities for future Mars exploration



The Mars Robotic Exploration Preparation Programme (MREP)

- **Mission studies for post-ExoMars missions**
 - PHOOTPRINT, INSPIRE, MPL-SFR, ..
- **Exploration Technology Development**
 - Mission specific technologies
 - Long-term enabling technologies (Nuclear Power Systems, Advanced Propulsion)
 - General MSR preparatory technology activities (eg. sampling, PP related technologies, sample rendezvous and capture, precision landing, ...)
- **MREP phase 2 ongoing, open for new subscriptions at C-MIN 2014**



Mars Sample Return: Current developments in ESA



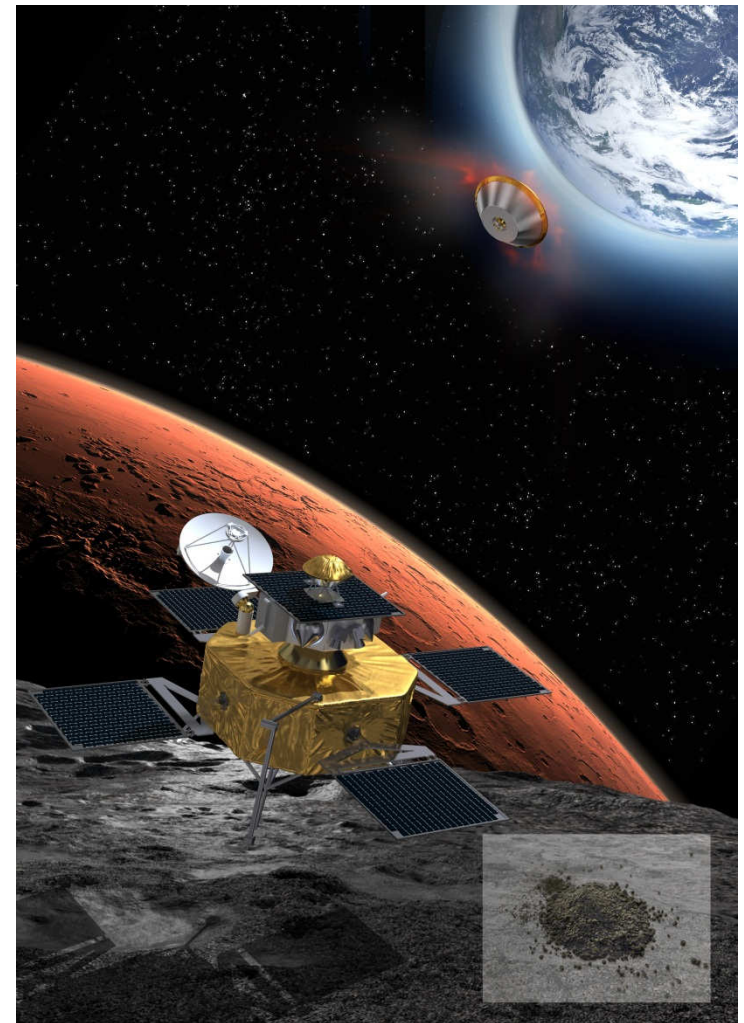
Technology development activities are implemented in ESA Mars Robotic Exploration Preparation Programme (MREP) and focused on specific themes:

- MSR Orbiter, including Orbiting Sample capture and biosealing in orbit, and Earth Return Capsule
- Sample fetch rover: Small rover (100 kg) with high surface mobility
- Precision landing on Mars: development of a capability to deliver 300 kg to Mars surface with a precision better than 10 km
- Sample receiving facility

PHOOTPRINT Mars-2024 candidate

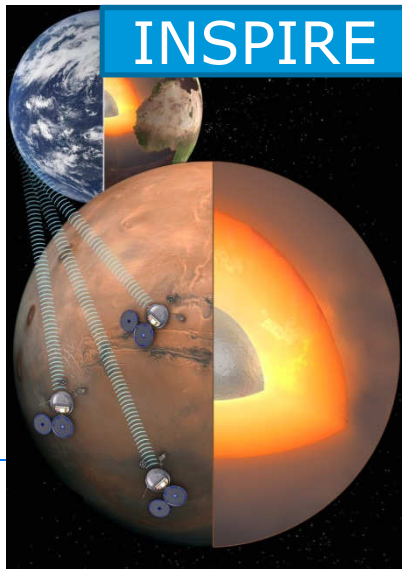


- **Return a sample of ~100 g from Phobos**
- **Reference Payload ~35 kg**
 - Cameras, VIS-NIR and Mid-IR spectrometers
 - Heritage from Rosetta, VEX and Bepi-Colombo
- **High and recognised intrinsic science value**
 - Origin and formation of Phobos; Solar System evolution
- **Prepares critical building blocks for MSR**
 - Sampling, transfer and sealing
 - Short range Rendezvous
 - Earth Return Capsule
 - Sample receiving facility
- **Possible scenarios for cooperation with Roscomos identified**
 - CDF study on joint mission on-going April-May 2014
 - Industrial phase A/B1 study in 2014 / 2015
 - Implementation proposal C-MIN 2016 (tbc)



Other candidate missions: INSPIRE and MPL

- Network of 3 surface elements for Mars interior and atmosphere characterisation.
- Timely after InSight and with potential of TGO being available
- Mature Payload, ~ 15 kg
- Seismometer, Mole with Heat Flow and Physical Properties Probe, Meteorological Boom, Radio-Science Package and Camera
- Heritage from ExoMars/Humboldt and Insight developments



- Soyuz launch of a Mars Precision Lander (MPL) + small science rover (10 km landing site accuracy)
- Explore a new region of Mars to investigate the (hydro-)geologic history, and habitability potential, taking benefit of high mobility
- Rover mobility > 170 m/sol capability, 3.5 to 5 hours available to drive per sol. GNC high mobility algorithms developed by MREP
- MSR preparation: MPL could be a segment of MSR Campaign, delivering the sample fetch rover

Back-up slides



NOMAD

High-resolution occultation and nadir spectrometers

Atmospheric composition
(CH_4 , O_3 , trace species, isotopes)
dust, clouds, P&T profiles

UVIS (0.20 – 0.65 μm) $\lambda/\Delta\lambda \sim 250$

SO Limb Nadir

IR (2.3 – 3.8 μm) $\lambda/\Delta\lambda \sim 10,000$

SO Limb Nadir

IR (2.3 – 4.3 μm) $\lambda/\Delta\lambda \sim 20,000$

SO



CaSSIS

High-resolution, stereo camera

Mapping of sources
Landing site selection



ACS

Suite of 3 high-resolution spectrometers

Atmospheric chemistry, aerosols,
surface T,
structure

Near IR (0.7 – 1.7 μm) $\lambda/\Delta\lambda \sim 20,000$

SO Limb Nadir

IR (Fourier, 2 – 25 μm) $\lambda/\Delta\lambda \sim 4000$ (SO)/500 (N)

SO Nadir

Mid IR (2.2 – 4.5 μm) $\lambda/\Delta\lambda \sim 50,000$

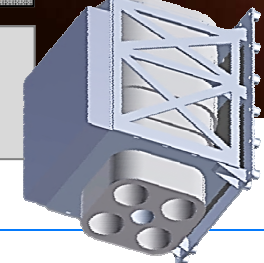
SO



FREND

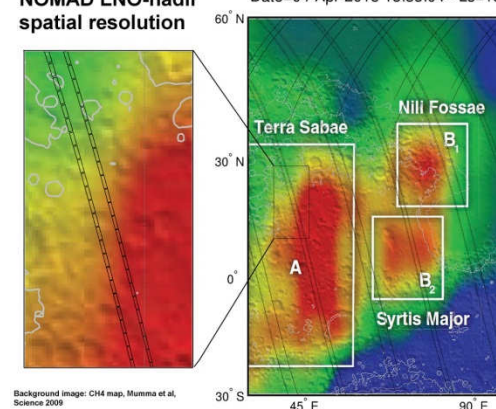
Collimated neutron detector

Mapping of subsurface water
And hydrated minerals



NOMAD LNO-nadir
spatial resolution

Date=04-Apr-2018 13:33:04 - Ls=154



Background image: CH4 map, Mumma et al., Science 2009

Credit: Kees Veenbos



PanCam

Wide-angle stereo camera pair
High-resolution camera

*Geological context
Rover traverse planning
Atmospheric studies*

WAC: 35° FOV, HRC: 5° FOV



ISEM

IR spectrometer on mast

*Bulk mineralogy of outcrops
Target selection*

$\lambda = 1.15 - 3.3 \mu\text{m}$, 1° FOV



CLUPI

Close-up imager

*Geological deposition environment
Microtexture of rocks
Morphological biomarkers*

20- μm resolution at 50-cm distance, focus: 20 cm to ∞



WISDOM

Ground-penetrating radar

*Mapping of subsurface
stratigraphy*

3 – 5-m penetration, 2-cm resolution



FREND

Passive neutron detector

*Mapping of subsurface
Water and hydrated minerals*



Drill + Ma_MISS

IR borehole spectrometer

In-situ mineralogy information

$\lambda = 0.4 - 2.2 \mu\text{m}$



Analytical Laboratory Drawer



MicrOmega

VIS + IR Spectrometer

*Mineralogical characterization
of crushed sample material
Pointing for other instruments*

$\lambda = 0.9 - 3.5 \mu\text{m}$, 256 x 256, 20- μm /pixel, 500 steps



RLS

Raman spectrometer

*Geochemical composition
Detection of organic pigments*

spectral shift range 200–3800 cm^{-1} , resolution $\leq 6 \text{ cm}^{-1}$



MOMA

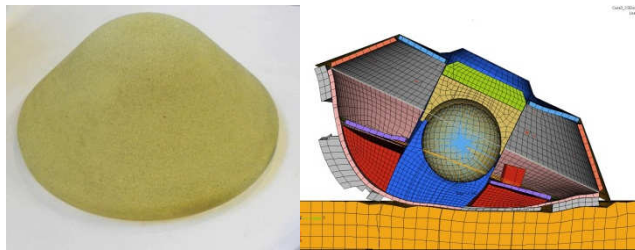
LDMS + Pyr-Dev GCMS

*Broad-range organic molecules
at high sensitivity (ppb)
Chirality determination*

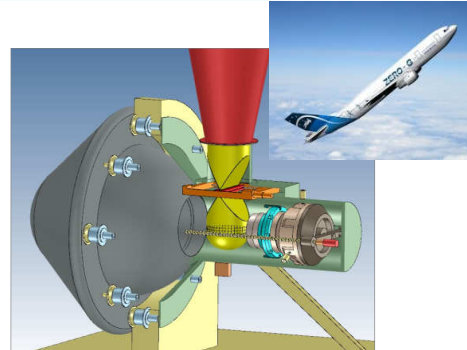
Laser-desorption extraction and mass spectroscopy

Pyrolysis extraction in the presence of derivatisation agents, coupled with chiral gas chromatography, and mass spectroscopy

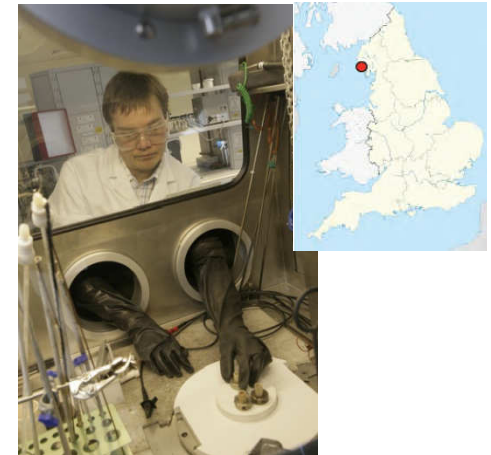
MREP Technology Development Overview



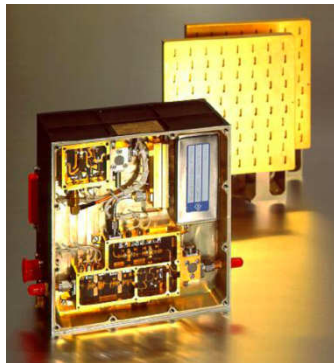
Earth Return Capsule developments



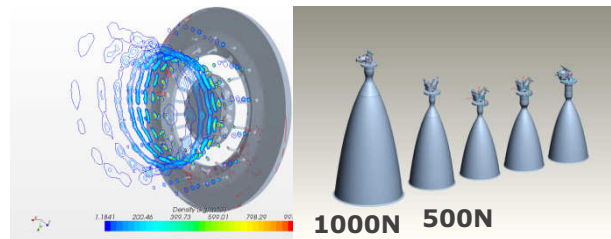
Sample Capture and biosealing in orbit developments



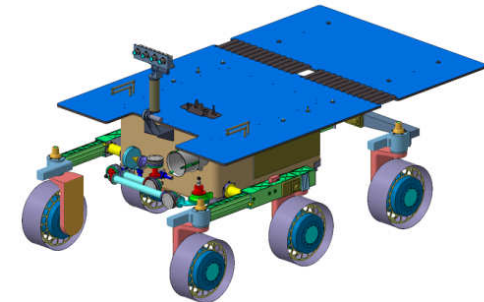
Nuclear Power Sources RHU & RPG , based on Am(241) radioisotope



Precision Landing equipment : Inertial Measurement Unit, altimeter miniaturisation



1 kN bi-liquid engine development for Mars insertion



Sample Fetch Rover and high mobility demonstration